

# Characterizing Solar Energetic Particles, from the Perspective of Engineering and Exploration

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Solar Energetic Particles, Solar Modulation and Space Radiation:
New opportunities in the AMS Era #3

**April 2018 Workshop** 

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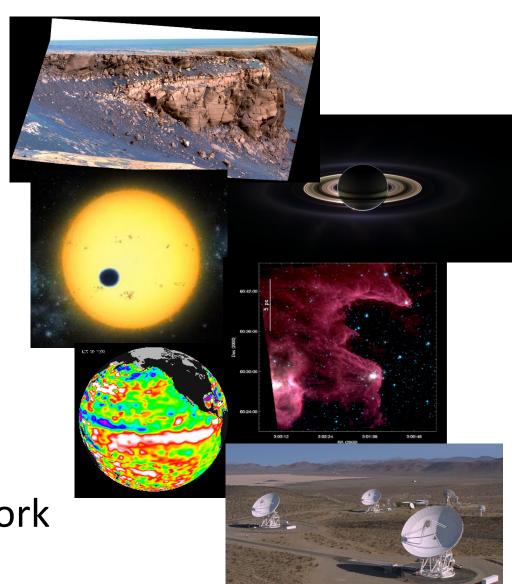
#### **Outline**

- Introduction to JPL and Natural Space Environments (NSE) Group
- Priorities for solar energetic proton characterization: Human vs. Robotic exploration
- Solar Proton Environment Specification for robotic Missions
- Need for further science contributions

# JPL's primary mission is robotic space exploration

- Solar system
- Exoplanets
- Astrophysics
- Earth Science

Interplanetary network



### Thirty-two spacecraft and instruments across the solar system (and beyond) – as of 2015



MSL

Wide-field Infrared Survey

**Explorer (WISE)** 

**Two Voyagers** 

SMAP

#### JPL's Natural Space Environment Group

- Supports all JPL space-flight missions for the space environments and effects
  - Radiation (environment, shielding, charging, ....),
  - Meteoroids, Debris (MMOD)
- Is the JPL lead for nuclear planetology (Gamma ray and neutron spectroscopy)
- Tasked with understanding environment hazards, to direct the design of robotic spacecraft we send into the solar system
- We strive to contribute to model development or improvement
  - e.g. models of solar proton fluence, outer-planet trapped radiation, meteoroids

# Interplanetary Radiation Issues: Robotic vs Human Exploration

#### GCR:

serious risk to astronauts negligible cumulative damage to electronic parts

-> solar-cycle variation matters to astronauts, not to parts

#### **SEP** events:

to be avoided by astronauts must be tolerated by electronics

#### -> SEP models:

- ·characterization of onset and flux profile is key so astronauts can hide
- characterization of cumulative exposure is key for part selection

Consequence: Some different aspects of the science are important in the two realms of Exploration.

### Solar Energetic Proton (SEP) Fluence contributes to the Mission Dose in amounts that depend on the mission.

Mission	SEP Dose (krad-Si)	Total Dose (krad-Si)	% from SEP
Moon Mission (for 1 yr)	2.68	2.68	100
Cassini (Saturn)	7.34	7.5	98
MSL (Mars rover)	1.46	3.41*	43
SMAP (Earth orbiter)	1.90	4.57	42
Europa Clipper (Jupiter)	5.69	3000	0.2

For 0.100" (~2.5 mm) aluminum shielding, Radiation design factor (RDF) = 1.

\* RTG contribution varies across rover.

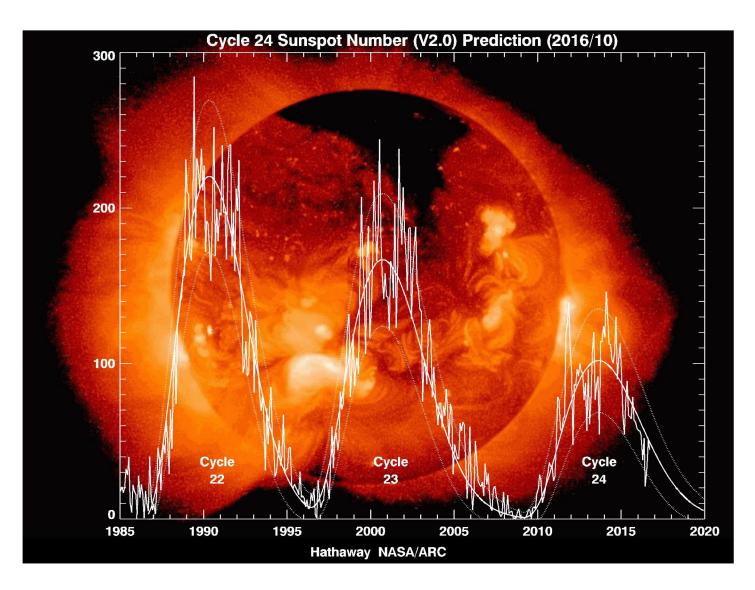
#### **Solar Proton Fluence Model**

Current class of models characterize proton fluence over X number of years at 1AU.

A dataset of observed events is fit to an assumed size distribution function of SEP events, and a frequency of events is determined, from which are derived fluence as a function of confidence level (i.e. probability that the fluence level will not to be exceeded).

Follows the lead set by King (1974).

# Amount of activity varies from cycle to cycle

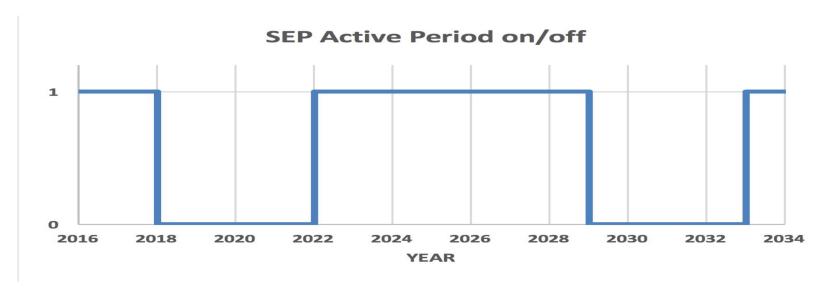


### Fluence Model is simple

Does not try to tie SEP activity to SSN peak or temporal variation except for bi-modal "active"/"quiet" periods.

Statistics apply to active period.

Assume ~ 4 years of quiet period around solar minimum.



# SEP Characterization for Robotic Exploration

 For probabilistic description of event size and frequency, we

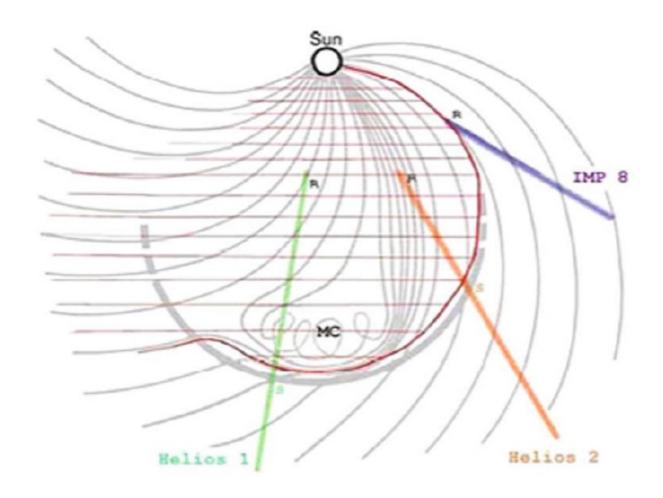
need continuous, long-timeline observations

For deep-space missions, we

#### need radial dependence of event fluence

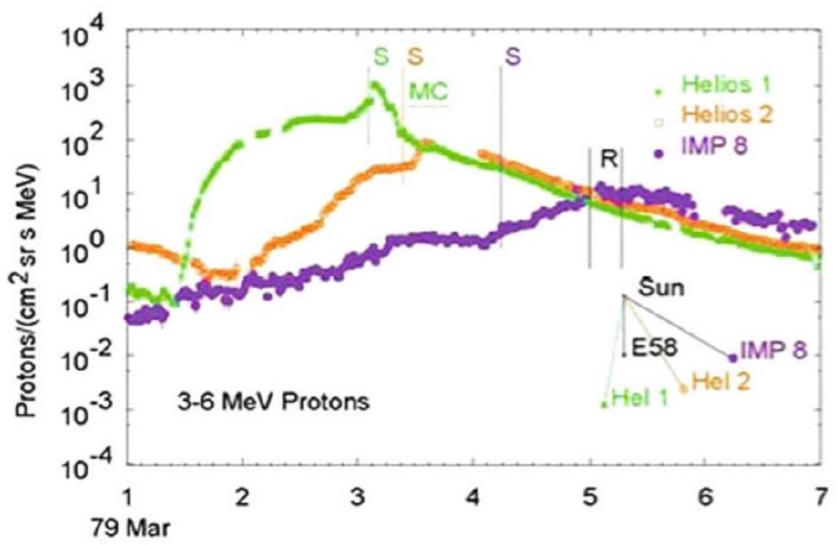
- Look at events during time periods when Earth and Mars are the same side of the sun
- look at event Size v. Frequency at different energy thresholds, and compare to statistics seen at 1AU

### Rare example of simultaneous multi-location measurement: locations



From D. Reames, Sp. Sci. Rev. (90), 1999.

## Rare example of simultaneous multi-location measurement: results



#### To Fill Knowledge Gap:

Scenario: Interplanetary space-weather buoy co-orbiting with Mars

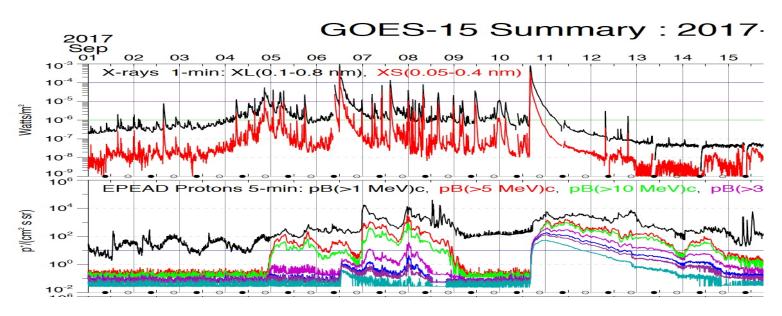
– e.g. spacecraft at Mars-Sun L1

 With e.g. GOES particle and x-ray spectrometers, Sun disk imaging (to correlate active regions to SEP events)

### To Fill Knowledge Gap:

#### Benefit:

- Characterize SEP environment at 1.5 AU (*i.e.*, Mars orbit)
- Enable long-timeline data sets to improve near-Mars & interplanetary weather and climate models



### There's still work to do (pick a curve)

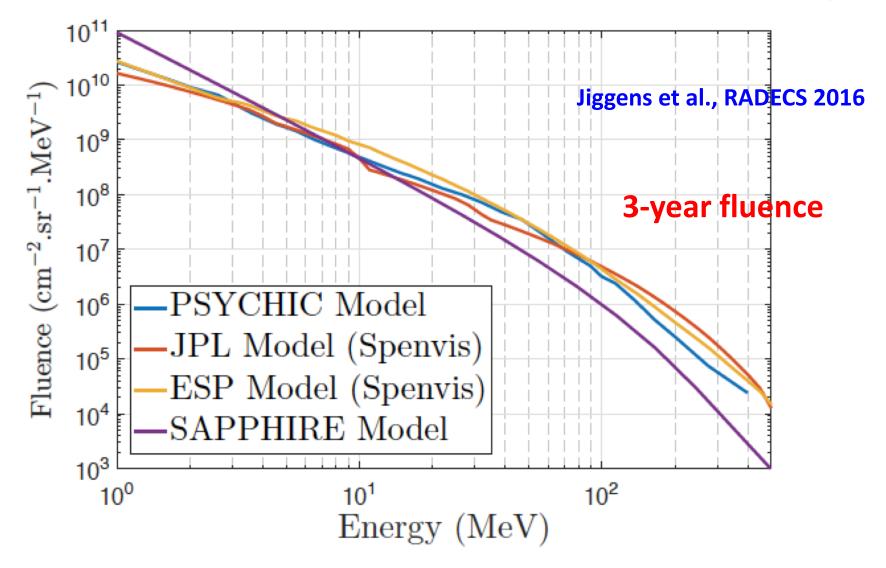


Fig. 5. Comaprison of cumulative fluence model outputs for protons at 95% confidence.

#### **Some References:**

- JPL model, also called the Feynman Model (Joan Feynman)
  - Feynman et al., JGR, 1993 (where it is called the JPL 1991 model)
  - Feynman et al., J. Atmos. Solar-Terr. Phys., 2002
- ESP model
  - Xapsos et al., IEEE Trans. Nucl. Sci. 2000
- http://test.sepem.eu/help/index\_intro.html
  - Solar Energetic Particle Environment Modelling (SEPEM)

